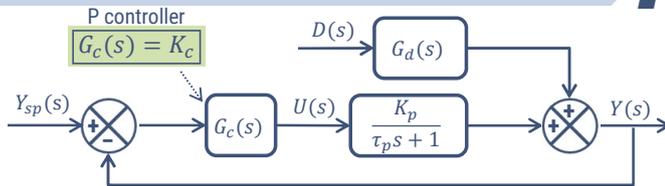




P controllers for FO «processes»: servo response



Overall process: FO system (at open loop)

G_p includes the overall loop dynamics and gains (except the controller ones)

$$Y = \frac{G_d}{1 + G_{OL}} D + \frac{K_m G_c G_v G_p}{1 + G_{OL}} Y_{sp}$$

$$CLTF_{sp} = \frac{Y(s)}{Y_{sp}(s)} = \frac{K_m G_c G_v G_p}{1 + G_c G_v G_p G_m} = \frac{K_c \frac{K_p}{\tau_p s + 1}}{1 + K_c \frac{K_p}{\tau_p s + 1}} = \frac{K_c K_p}{\tau_p s + 1 + K_c K_p} = \frac{K_c K_p}{\tau_p s + (1 + K_c K_p)} = \frac{K_c K_p}{\frac{\tau_p}{1 + K_c K_p} s + 1}$$

$$CLTF_{sp}: \frac{Y(s)}{Y_{sp}(s)} = \frac{K_1}{\tau_1 s + 1} \begin{cases} K_1 = \frac{K_c K_p}{1 + K_c K_p} = \frac{K_{OL}}{1 + K_{OL}} < 1 \\ \tau_1 = \frac{\tau_p}{1 + K_c K_p} = \frac{\tau_p}{1 + K_{OL}} < \tau_p \end{cases}$$

$K_{OL} > 0$: gain of the open loop

Remarks:

- ❖ If the loop is closed with a P controller, the closed-loop response is still a **FO response**
- ❖ Since $K_1 < 1$ (hence $\neq 1$), y can never attain y_{sp} → **offset**
- ❖ Since $\tau_1 < \tau_p$, the closed-loop response is **faster** than the open-loop one

Adapted from Barolo M.

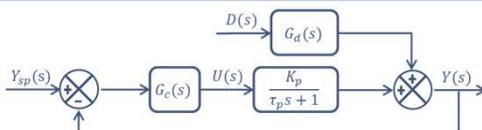
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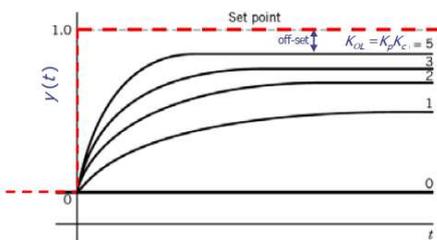
P controllers for FO «processes»: servo response /cont'd



$$CLTF_{sp}: \frac{Y(s)}{Y_{sp}(s)} = \frac{K_1}{\tau_1 s + 1} \begin{cases} K_1 = \frac{K_{OL}}{1 + K_{OL}} < 1 \\ \tau_1 = \frac{\tau_p}{1 + K_{OL}} < \tau_p \end{cases}$$

P controller
 $G_c(s) = K_c$

Servo problem: $Y_{sp} = \frac{1}{s}$ (unit step)



(Smith and Corripio, 2005)

Off-set:

$$y_{sp} - y(\infty) \rightarrow y_{sp} - \lim_{s \rightarrow 0} \left[s \left(\frac{K_1}{s(\tau_1 s + 1)} \right) \right] = 1 - K_1 = \frac{1}{1 + K_{OL}}$$

- ❖ If $K_c > 0$ increases, **the dynamics is made faster** and the off-set decreases
- ❖ What happens if $K_{OL} < 0$?

Adapted from Barolo M.

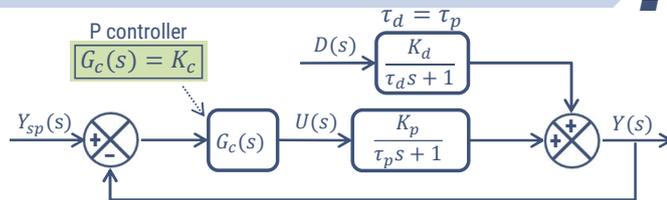
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P controllers for FO «processes»: disturbance rejection



Overall process: FO system (at open loop)

G_p includes the overall loop dynamics and gains (except the controller ones)

$$Y = \frac{G_d}{1 + G_{OL}} D + \frac{K_m G_s G_v G_p}{1 + G_{OL}} Y_{sp}$$

$$CLTF_{sp} = \frac{Y(s)}{D(s)} = \frac{G_d}{1 + G_c G_v G_p G_m} = \frac{\frac{K_d}{\tau_p s + 1}}{1 + K_c \frac{K_p}{\tau_p s + 1}} = \frac{\frac{K_d}{\tau_p s + 1}}{\frac{\tau_p s + 1 + K_c K_p}{\tau_p s + 1}} = \frac{K_d}{\tau_p s + (1 + K_c K_p)} = \frac{\frac{K_d}{1 + K_c K_p}}{\frac{\tau_p}{1 + K_c K_p} s + 1}$$

$$CLTF_d: \frac{Y(s)}{D(s)} = \frac{K_2}{\tau_1 s + 1} \begin{cases} K_2 = \frac{K_d}{1 + K_c K_p} = \frac{K_d}{1 + K_{OL}} > 0 \\ \tau_1 = \frac{\tau_p}{1 + K_c K_p} = \frac{\tau_p}{1 + K_{OL}} < \tau_p \end{cases}$$

$K_{OL} > 0$: gain of the open loop

Remarks:

- ❖ If the loop is closed with a P controller, the closed-loop response is still a **FO response**
- ❖ Since $K_2 > 0$ (hence $\neq 0$), y can be **never** restored back to y_{sp} at steady state \rightarrow **offset**
- ❖ Since $\tau_1 < \tau_p$, the closed-loop response is **faster** than the open-loop one

Adapted from Barolo M.

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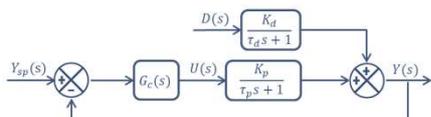
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P controllers for FO «processes»: disturbance rejection

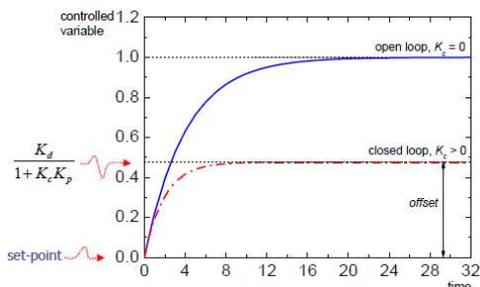
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$$CLTF_d: \frac{Y(s)}{D(s)} = \frac{K_2}{\tau_1 s + 1} \begin{cases} K_2 = \frac{K_d}{1 + K_{OL}} > 0 \\ \tau_1 = \frac{\tau_p}{1 + K_{OL}} < \tau_p \end{cases}$$

$$\text{P controller } G_c(s) = K_c$$

Disturbance rejection: $D(s) = \frac{1}{s}$ (unit step)



Off-set:

$$y_{sp} - y(\infty) \rightarrow y_{sp} - \lim_{s \rightarrow 0} \left[s \left(\frac{K_2}{s(\tau_1 s + 1)} \right) \right] = 0 - K_2 = \frac{-K_d}{1 + K_c K_p}$$

- ❖ If $K_c > 0$ increases, **the dynamics is made faster** and the off-set decreases

Adapted from Barolo M.

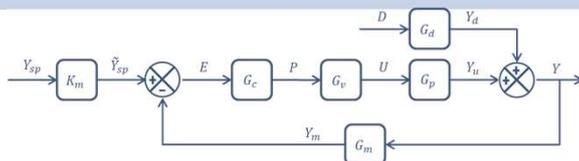
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GENERALIZING CLTFs UNDER P CONTROL



P controller
 $G_c(s) = K_c$

If the **only system dynamics** are those of the process and of the disturbance, and if they all are FO ($K_p; K_d; \tau_p$), then the CLTFs for a P-only controller are (for a step change of amplitude M):

Servo control: $CLTF_{sp}: \frac{Y(s)}{Y_{sp}(s)} = \frac{K_1}{\tau_1 s + 1} \left\{ \begin{array}{l} K_1 = \frac{K_{OL}}{1 + K_{OL}} < 1 \\ \tau_1 = \frac{\tau_p}{1 + K_{OL}} < \tau_p \end{array} \right.$ Offset: $\frac{M}{1 + K_{OL}}$

The dynamics is still first-order, but it is made **faster**

Regulatory control: $CLTF_d: \frac{Y(s)}{D(s)} = \frac{K_2}{\tau_1 s + 1} \left\{ \begin{array}{l} K_2 = \frac{K_d}{1 + K_{OL}} > 0 \\ \tau_1 = \frac{\tau_p}{1 + K_{OL}} < \tau_p \end{array} \right.$ Offset: $\frac{-K_d M}{1 + K_{OL}}$

Adapted from Barolo M.

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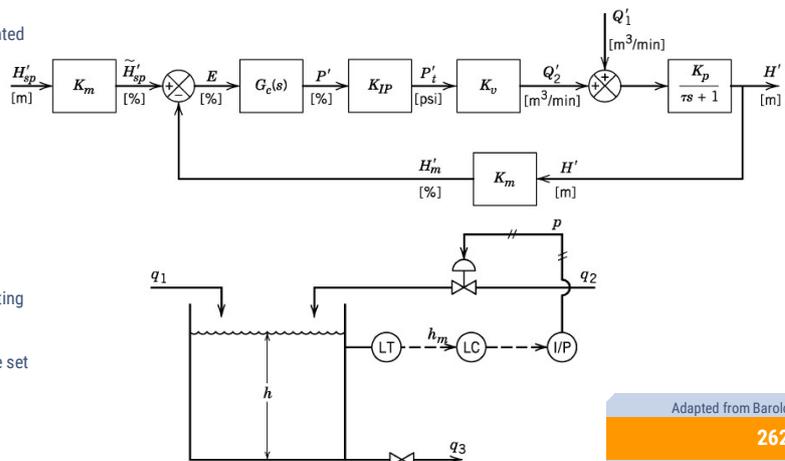
EXAMPLES



Examples: SEMD 4° Examples 11.2 + 11.3 p.183-184; 3° Examples 11.2 + 11.3 p.191-192

Consider the level control system shown in the figure, implemented with a computer whose inputs and outputs are calibrated in terms of full range (100%). The tank is 1 m in diameter, while the valve on the exit line acts as a linear resistance with $R = 6.37 \text{ min/m}^2$. The level transmitter has a span of 0.5 m and an output range of 0 to 100%. The valve characteristic f of the equal percentage control valve is related to the fraction of lift ℓ by the relation $f(\ell) = (30)^\ell - 1$. The air-to-open control valve receives a 3 to 15 psi signal from an I/P transducer, which, in turn, receives a 0 to 100% signal from the computer-implemented proportional-only controller. When the control valve is fully open ($f(\ell) = 1$), the flow rate through the valve is $0.2 \text{ m}^3/\text{min}$. At the nominal operating condition, the control valve is half-open ($f(\ell) = 0.5$).

Calculate the closed-loop responses to a unit step change in the set point for three values of the controller gain: $K_c = 1, 2, \text{ and } 5$.



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PI controllers for FO «processes»: servo response

PI controller $G_c(s) = K_c \left(1 + \frac{1}{\tau_I s} \right)$

Overall process: FO system (at open loop)

G_p includes the overall loop dynamics and gains (except the controller ones)

$$Y = \frac{G_d}{1 + G_{OL}} D + \frac{K_m G_c G_v G_p}{1 + G_{OL}} Y_{sp}$$

$$CLTF_{sp} = \frac{Y(s)}{Y_{sp}(s)} = \frac{K_m G_c G_v G_p}{1 + G_c G_v G_p G_m} = \frac{\left(K_c \frac{\tau_I s + 1}{\tau_I s} \right) \frac{K_p}{\tau_p s + 1}}{1 + \left(K_c \frac{\tau_I s + 1}{\tau_I s} \right) \frac{K_p}{\tau_p s + 1}} = \frac{\frac{K_c K_p (\tau_I s + 1)}{\tau_I s (\tau_p s + 1)}}{(\tau_I \tau_p s^2 + \tau_I s) + K_c K_p (\tau_I s + 1)} = \frac{K_c K_p (\tau_I s + 1)}{\tau_I \tau_p s^2 + \tau_I (1 + K_c K_p) s + K_c K_p}$$

Remarks:

- ❖ If the loop is closed with a PI controller, the closed-loop response has a **FO numerator** and a **SO denominator** → the dynamics of the closed-loop system becomes SO with numerator dynamics
- ❖ The net effect of P-only control was instead to only change the system parameters, while keeping FO the response
- ❖ A PI controller, instead, **adds a pole and a zero** to the closed-loop system response (for a FO system)

Adapted from Barolo M. 263

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PI controllers for FO «processes»: servo response /cont'd

PI controller

$$G_c(s) = K_c \left(1 + \frac{1}{\tau_I s} \right)$$

$$CLTF_{sp} = \frac{Y(s)}{Y_{sp}(s)} = \frac{K_c K_p (\tau_I s + 1)}{\tau_I \tau_p s^2 + \tau_I (1 + K_c K_p) s + K_c K_p} = \frac{\tau_I s + 1}{\frac{\tau_I \tau_p}{K_c K_p} s^2 + \frac{\tau_I (1 + K_c K_p)}{K_c K_p} s + 1} = \frac{\tau_I s + 1}{\tau^2 s^2 + 2\zeta \tau s + 1}$$

$\tau = \sqrt{\frac{\tau_I \tau_p}{K_c K_p}} = \sqrt{\frac{\tau_I \tau_p}{K_{OL}}}$

$2\zeta \tau = \frac{\tau_I (1 + K_c K_p)}{K_c K_p} \rightarrow \zeta = \frac{1}{2} (1 + K_c K_p) \sqrt{\frac{\tau_I}{\tau_p K_c K_p}} = \frac{1}{2} \left(\frac{1 + K_{OL}}{\sqrt{K_{OL}}} \right) \sqrt{\frac{\tau_I}{\tau_p}}$

■ For $Y_{sp} = M/s$, the response is of this type: $y(t) = A_0 + A_1 e^{p_1 t} + A_2 e^{p_2 t}$

- ▷ The numerical values of the roots p_i depend on the **process** parameters **and** on the **controller** parameters

Remarks:

- ❖ Since the CLTF has one zero, the closed-loop response **may** display an overshoot
- ❖ If the two roots p_1 and p_2 are **real and negative**, the response exponentially attains a **final value A_0**
- ❖ If the two roots p_1 and p_2 are **complex conjugate**, and with **negative real parts**, the response will be **oscillating**, but the oscillations eventually die out (**underdamped response**)
- ❖ If at least **one** of the roots is **real and positive** (or the roots are complex with positive real part), then the response will not settle to a final value, but will **diverge** (possibly with oscillations)

Adapted from Barolo M. 264

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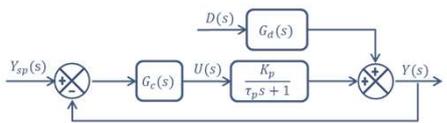


PI controllers for FO «processes» - Removing the offset

PI controller

$$G_c(s) = K_c \left(1 + \frac{1}{\tau_I s} \right)$$

$$CLTF_{sp} = \frac{Y(s)}{Y_{sp}(s)} = \frac{\tau_I s + 1}{\tau^2 s^2 + 2\zeta\tau s + 1} \quad Y_{sp}(s) = \frac{M}{s}$$



What will be the final response value?

$$y(\infty) = \lim_{s \rightarrow 0} [sY(s)] = \lim_{s \rightarrow 0} \left[s \left(\frac{M}{s} \times \frac{\tau_I s + 1}{\tau^2 s^2 + 2\zeta\tau s + 1} \right) \right] = M$$

The final value is equal to the new set-point M



- The integral action can eliminate the offset in servo control problems
 - For disturbance rejection problems, the result is similar

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PI controllers for FO «processes» - Removing the offset

/cont'd

PI controller

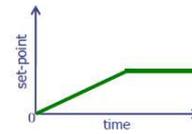
$$G_c(s) = K_c \left(1 + \frac{1}{\tau_I s} \right)$$

- The integral action can completely compensate for load (or set-point) changes, provided that the change is sustained
 - For example, the integral action cannot compensate for a ramp-like change (of a disturbance or set-point) during the ramp evolution
 - ❖ But when the transient has died out, the change can be completely compensated

$$CLTF_{sp} = \frac{Y(s)}{Y_{sp}(s)} = \frac{\tau_I s + 1}{\tau^2 s^2 + 2\zeta\tau s + 1}$$

$$\tau = \sqrt{\frac{\tau_I \tau_p}{K_c K_p}} = \sqrt{\frac{\tau_I \tau_p}{K_{OL}}}$$





$$\zeta = \frac{1}{2} (1 + K_c K_p) \sqrt{\frac{\tau_I}{\tau_p K_c K_p}} = \frac{1}{2} \left(\frac{1 + K_{OL}}{\sqrt{K_{OL}}} \right) \sqrt{\frac{\tau_I}{\tau_p}}$$

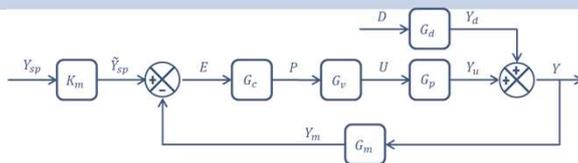
- Effect of K_c and τ_I
 - If $K_c \uparrow$ or $\tau_I \downarrow$ the characteristic time $\tau \downarrow$: the response is sped up
 - If $\tau_I \downarrow$, the damping factor $\zeta \downarrow$: the response becomes more oscillatory (i.e., less stable)
 - It may seem that the same happens also if $K_c \downarrow$
 - ❖ But this is due to the fact that the valve and transmitter dynamics have been neglected; if they are accounted for, the response becomes more oscillatory when $K_c \uparrow$ (as always happens in practice)

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GENERALIZING CLTFs UNDER PI CONTROL



PI controller

$$G_c(s) = K_c \left(1 + \frac{1}{\tau_I s} \right)$$

- Similar consideration hold true also for systems of order greater than 1
- The **integral action modifies the order of the system response**
- For any dynamics system $G_p(s)$ with **steady-state gain K_p** , the following holds true:
 - The closed-loop response under **proportional-only** feedback control **always shows offset**
 - The **integral action can eliminate the steady-state offset** from the closed-loop response, provided that the disturbance or set-point change are sustained
 - **Increasing the controller gain K_c** makes the response **more oscillatory**

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STABILITY OF A CONTROLLED SYSTEM

An unconstrained linear system is said to be **stable** if the output response is bounded for all bounded inputs. Otherwise, it is said to be **unstable**

- Most physical systems are **open-loop stable** (self-regulating systems)
 - They attain a new steady state when a sustained input change is applied
 - They are «inherently» stable, i.e., they are stable when they are not part of a feedback control loop
- Some real processes are **open-loop unstable**
 - To be operated, they need a feedback control loop that stabilizes them
 - ❖ Example: some levels (if pure integrators); exothermic reactors with cooling jacket (generated heat vs removed heat)
- Even if a process is **stable at open loop, not necessarily** does it remain stable when the control loop is **closed**

- ❖ Unconstrained system or «**sistema non vincolato**»: there are no physical limits on the input and output variables
- ❖ Bounded signal or «**segnale confinato**»: the signal remains within upper and lower limits for all values of time

Adapted from Barolo M.

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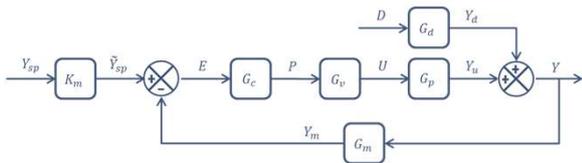
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EFFECT OF THE CONTROLLER ON THE RESPONSE

Example: SEMD 4° Ex.11.4 p. 187; 3° Ex. 11.4 p. 195



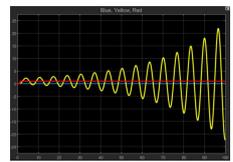
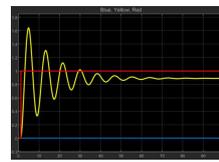
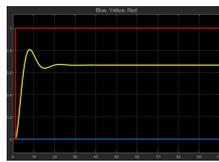
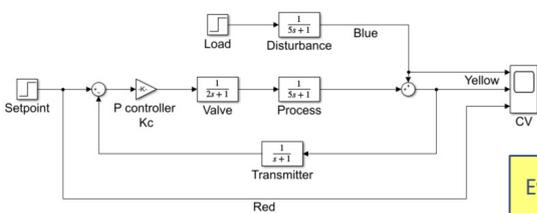
$$G_p = G_d = \frac{1}{5s + 1} \quad G_c = K_c$$

$$G_v = \frac{1}{2s + 1} \quad G_m = \frac{1}{s + 1}$$

$K_c = 2$

$K_c = 8$

$K_c = 15$



Even if P-only control is used, there exists a limiting value of K_{cu} of the controller gain (ultimate gain or «guadagno critico») beyond which the response becomes unstable

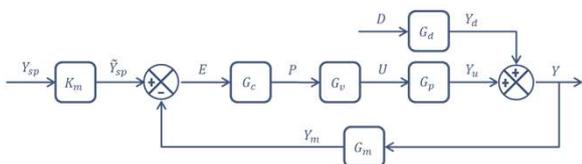
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EFFECT OF THE CONTROLLER ON THE RESPONSE /cont'd

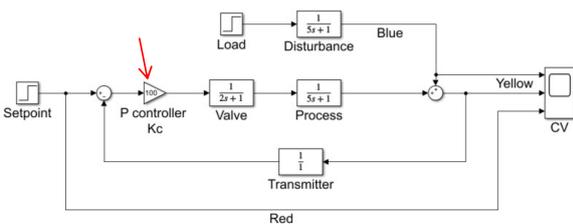
Example: SEMD 4° Ex.11.4 p. 187; 3° Ex. 11.4 p. 195



$$G_p = G_d = \frac{1}{5s + 1} \quad G_c = K_c$$

$$G_v = \frac{1}{2s + 1} \quad G_m = 1 \neq \frac{1}{s + 1}$$

$K_c = 100$



It looks like that a limiting value for the controller gain does not exist for all processes

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SIGNAL AMPLIFICATION WITHIN THE LOOP

- For some systems, the **signal variation reinforce** (amplify) when travel around the loop, **even if P-only control is used**
 - ▷ The amplifications may be such that the **oscillations** become unbounded even for a **bounded** input change
 - ❖ In reality, as a result of physical constraints (example: valve opening/closure; structural resistance of an equipment) the oscillations amplitude will be bounded and/or there will be structural failure of the equipment
- To design the process control system:
 - ▷ We must know whether or not an ultimate gain value exists for a given system
 - ▷ When it exists, we must be able to calculate it

Adapted from Barolo M.

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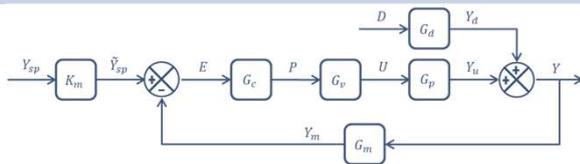
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CHARACTERISTIC EQUATION

$$1 + G_{OL} = 0$$



$$G_{OL} \triangleq G_c G_v G_p G_m$$

$$Y = \frac{G_d}{1 + G_{OL}} D + \frac{K_m G_c G_v G_p}{1 + G_{OL}} Y_{sp}$$

As an example, consider set-point changes only: $D(s) = 0$

If no repeated poles exist, the CLTF can be factored as:

$$\frac{Y(s)}{Y_{sp}(s)} = \frac{K_m G_c G_v G_p}{1 + G_{OL}} = \frac{b_m(s - z_1)(s - z_2) \dots (s - z_m)}{a_n(s - p_1)(s - p_2) \dots (s - p_n)} \rightarrow$$

$$Y(s) = \frac{\text{numerator terms}}{a_n(s - p_1)(s - p_2) \dots (s - p_n) (\text{input terms})} \rightarrow$$

The input is bounded

$$Y(s) = \frac{A_1}{(s - p_1)} + \frac{A_2}{(s - p_2)} + \dots + \frac{A_n}{(s - p_n)} + (\text{addendum from input terms})$$

Notice.
The poles p_k are the solutions of the characteristic equation of the closed-loop system: $1 + G_{OL} = 0$

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UNFORCED RESPONSE AND FORCED RESPONSE

$$Y(s) = \frac{A_1}{(s - p_1)} + \frac{A_2}{(s - p_2)} + \dots + \frac{A_n}{(s - p_n)} + (\text{addendum from input terms})$$

$$y(t) = \underbrace{A_1 e^{p_1 t} + A_2 e^{p_2 t} + \dots + A_n e^{p_n t}}_{\text{Unforced response}} + \underbrace{(\text{addendum from input terms})}_{\text{Forced response}}$$

- Each term of the **unforced response** includes a root (**pole**) of the characteristic equation (closed-root response)
 - ▷ Since the values of the A_k coefficients depend on the actual forcing input (and on the zeroes of the CLTF), the response **shape** $y(t)$ does **depend on the input**
 - ▷ However, the speed at which $y(t)$ decays or diverges is determined by the **roots of the closed-loop characteristic equation only**

$$1 + G_{OL} = 0$$

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STABILITY CONDITIONS

- Closed-loop response:

$$y(t) = A_1 e^{p_1 t} + A_2 e^{p_2 t} + \dots + A_n e^{p_n t} + (\text{addendum from input terms})$$

- **Real poles p_k**

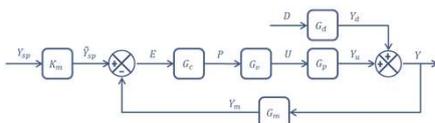
- ▷ **Positive**

- ❖ If for any closed-loop pole p_k it results that $p_k > 0 \rightarrow e^{p_k t}$ becomes unbounded $\rightarrow y(t)$ becomes unbounded \rightarrow **the overall response is unstable**

- **Complex poles p_k**

- ▷ **Positive real part**

- ❖ For **any** pole $p_k = \sigma \pm j\omega$, it follows that $e^{p_k t} = e^{(\sigma \pm j\omega)t} = e^{\sigma t} \cdot e^{\pm j\omega t}$
- ❖ If $\sigma > 0 \rightarrow e^{\sigma t}$ becomes unbounded \rightarrow **the overall response is unbounded** (and also oscillating as a result of $e^{\pm j\omega t}$)



A feedback control system is **stable** if and only if **all** roots of the closed-loop characteristic equation are **negative** (if real) or have **negative real parts** (if complex).
Otherwise, the system is **unstable**

Barolo M.

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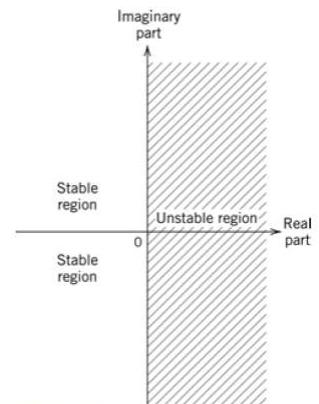
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REPRESENTATION IN THE COMPLEX PLANE

$1 + G_{OL} = 0$

- The solutions of the closed-loop characteristic equation that lie in the **right-half plane** give rise to **instability conditions** in the controlled system
 - ▷ Notice: the poles location depends on G_c !
- The root **positions** with respect to the imaginary axis indicate how **rapid** the transient response will be
 - ▷ The (real) solutions $s = a < 0$ corresponds to a **closed-loop** time constant $\tau = -1/a$
 - ▷ The farther **left** the root with respect to the imaginary axis, the **faster** the closed-loop response (small τ)
 - ▷ With reference to time, the **dominant responses** (slow) are those corresponding to **poles close to the imaginary axis**.
- The farther away the (complex) solutions from the **real axis**, the **more oscillatory** the transient response



(Seborg et al., 2011)

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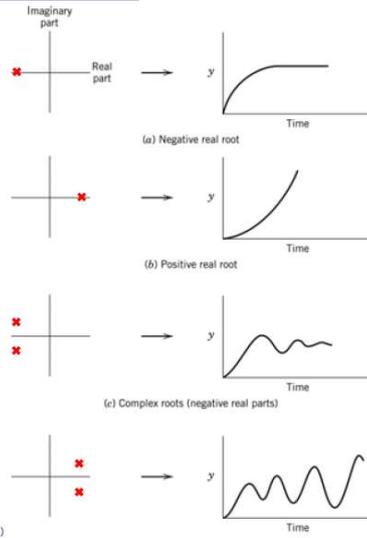
ROOT LOCATIONS AND CLOSED-LOOP RESPONSE

$1 + G_{OL} = 0$

Note that the dynamic elements that are **outside the loop** have **no effect on the stability** of the controlled system!

Final remarks:

- The stability criterion also holds for:
 - ▷ Load disturbances
 - ▷ Repeated poles
- If nonlinear elements (e.g., dead times) are included within G_{OL} , they need to be approximated (e.g., Padè)
 - ▷ Alternatively, a rigorous stability condition can be derived via frequency analysis (SEMD, Chapter 14)



(Seborg et al., 2011)

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