

# Leghe del TITANIO

Il titanio puro e' polimorfo esistendo in due strutture:

il titanio  $\alpha$  a reticolo esagonale compatto stabile fino a 882 C

il titanio  $\beta$ , cubico a corpo centratoo stabile fra 882 C e la T di fusione che e' di 1678 C.

Le proprieta' del titanio a temperatura ambiente sono molto interessanti

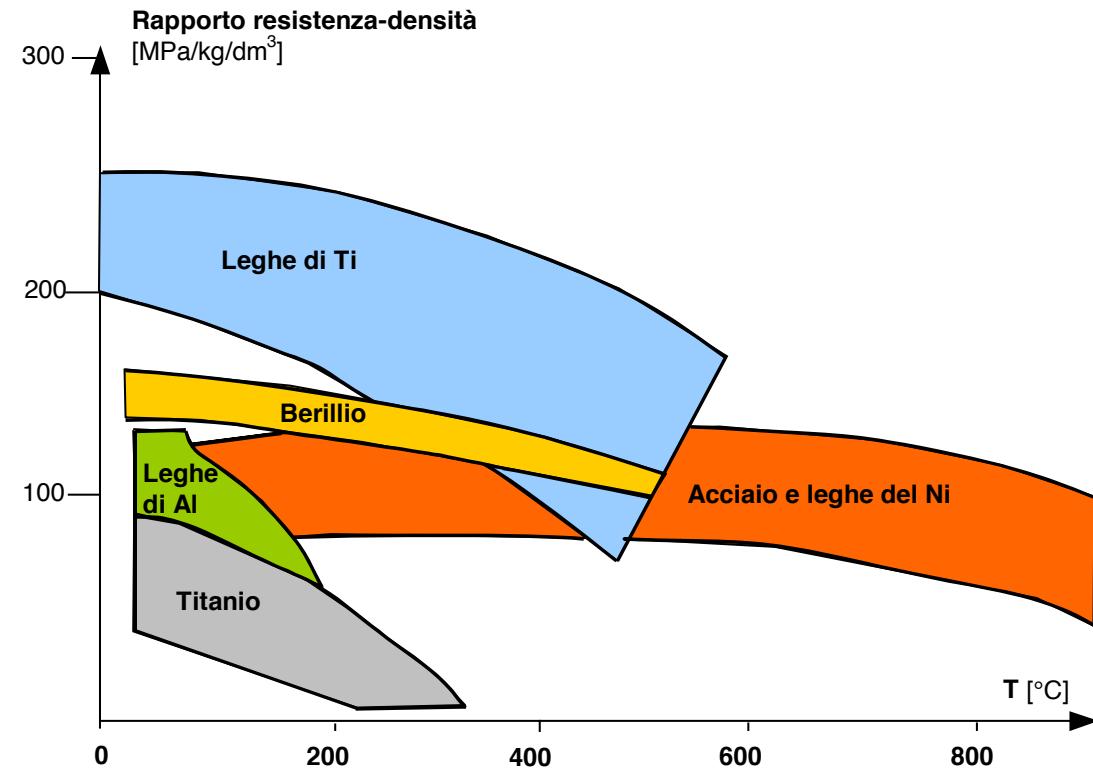
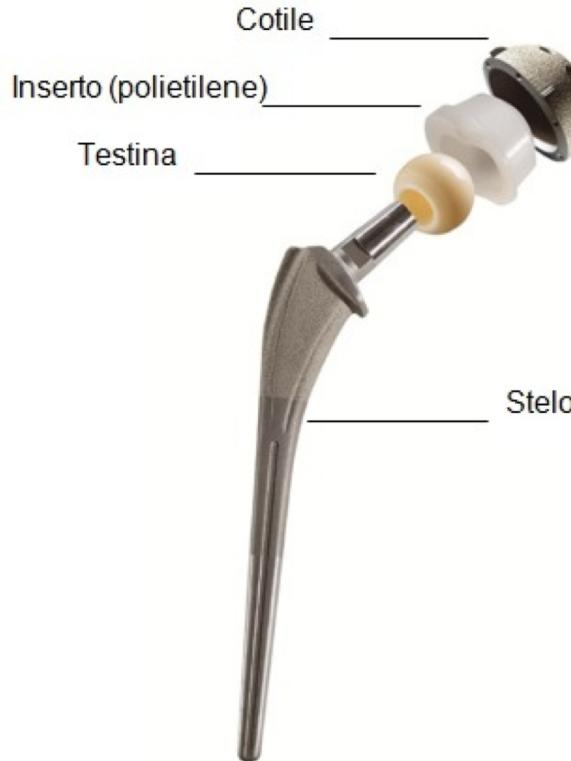
- densita' = 4,51 g/cm<sup>3</sup>
- $\sigma_y = 460 \text{ MPa}$  e  $\sigma_T = 560 \text{ MPa}$   $\varepsilon = 22\%$ , **HB = 200 - 220 (bassa!)**
- se viene laminato  $\sigma_T = 800 \text{ MPa}$  e  $\varepsilon = 8\%$
- modulo di Young = 114 GPa
- non e' magnetico
- conducibilita' termica = 19 W/m C
- coefficiente di espans. termica =  $8,4 \times 10^{-6} \text{ C}$

Dal punto di vista della resistenza alla corrosione il titanio ha un ottimo comportamento in molti ambienti paragonabile a quello di un acciaio inossidabile 18/8. La resistenza del titanio all'acqua di mare e' paragonabile a quella del platino, la resistenza all'erosione e' anche altissima (acqua di mare ad alta velocita' e vapore in pressione), 20 volte superiore a quella delle leghe Cu-Ni.

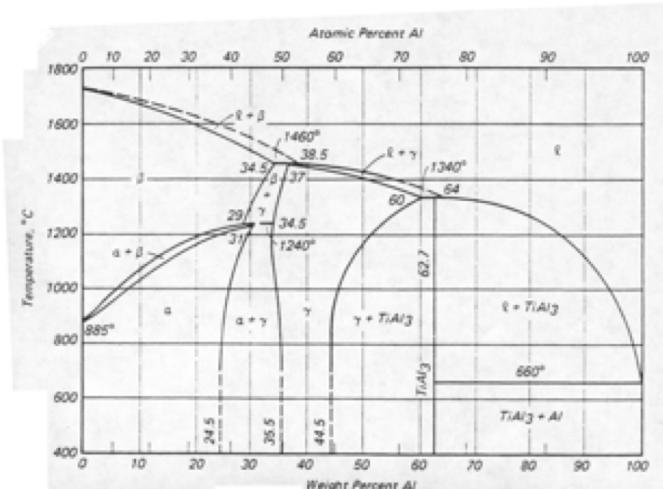
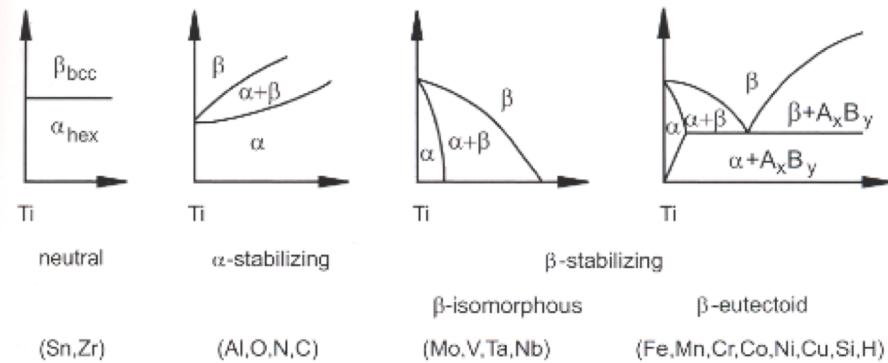
All'incirca il 95% del titanio viene consumato in forma di **ossido** ( $TiO_2$ ), nelle **vernici**, nella **carta**, nei **cementi** per renderli più brillanti e nelle **plastiche**. Le vernici fatte con il biossido di titanio riflettono molto bene la **radiazione infrarossa** e sono quindi molto usate dagli astronomi.

Il consumo mondiale di Titanio (2007) e' stato di circa 125.000 ton

- aeronautico e difesa (48.000 ton)
- chimico, petrolchimico, energetico, edile (48.000 ton)
- beni di consumo (10.000 ton)
- biomedico



A temperatura ambiente il Ti ha struttura cristallina **esagonale chiamata alfa**; a circa 882°C, la fase alfa si trasforma in una struttura **cubica a corpo centrato, chiamata beta**, che è stabile fino al punto di fusione 1660 °C.



## Leghe industriali del titanio

La principale influenza degli elementi aggiunti al titanio è quella di allargare o di restringere i campi di esistenza della fase  $\alpha$ .

1 = **Al, O, N, C, Ga** ( $\alpha$  stabilizzanti)

2 = **Mo, W, V, Ta** ( $\beta$  stabilizzanti)

3 = (diagrammi con eutettico), Cu, Mn, Cr, Fe, Ni, Co, H

----- = inizio e fine trasformazione martensitica

Tra gli elementi stabilizzanti della fase  $\alpha$  si ricordano principalmente **Al, O, N**

Questi elementi hanno inoltre la proprietà di produrre un indurimento della fase solida  $\alpha$ .

stabilizzanti della fase  $\beta$  sono invece altri elementi quali il **V, Mn, Cr, Fe, Nb, Mo, Ta**.

I due più importanti sono il **V** ed il **Mo**. Il Mo è poco usato, poiché ha un'alta densità.

## Leghe $\alpha$

Le leghe alfa sono un pò meno resistenti a corrosione rispetto al titanio non legato, ma possiedono resistenza maggiore sia di quest'ultimo che di tutte le altre leghe di titanio; inoltre, resistono all'ossidazione ad alte temperature ( $300^{\circ}\text{C}$  -  $540^{\circ}\text{C}$ ) e presentano miglior saldabilità rispetto ai vari tipi di titanio CP ed ottima duttilità.

*I livelli di resistenza a temperatura ambiente, comunque, sono i più bassi e queste composizioni non rispondono a trattamento termico.*

La principale variabile microstrutturale delle leghe alfa è la dimensione del grano.

Vengono divise in tre gruppi:

- solamente fase  $\alpha$
- fino a 2% di fase  $\beta$  (**near  $\alpha$** )
- invecchiabili (fino a 2.5% di Cu) - **Ti<sub>2</sub>Cu**

La lega piu' nota del primo gruppo e' la  
**Ti - 5Al - 2.5Sn (adesso anche ELI)**  
 $\sigma = 850 \text{ MPa}$ . (puo' operare anche a -200 C)



Una tipica lega **near  $\alpha$**  e' la  
**Ti - 11Sn - 2.25Al - 5Zr - 1Mo - 0,2Si**

Lo **Sn** e' migliore dell' Al (s.s.) ad alte temperature

**Zr** da anche s.s.

**Mo** stabilizza la fase  $\beta$

**Si** si scioglie nella fase  $\alpha$  e aumenta la resistenza meccanica e quella al creep

Sono state messe in commercio nuove composizioni con **0.2% di Pd** per migliorare la resistenza alla corrosione (1500 volte meglio) e altre composizioni chiamate **ELI** per migliorare la tenacita'.

## Beta titanium alloys

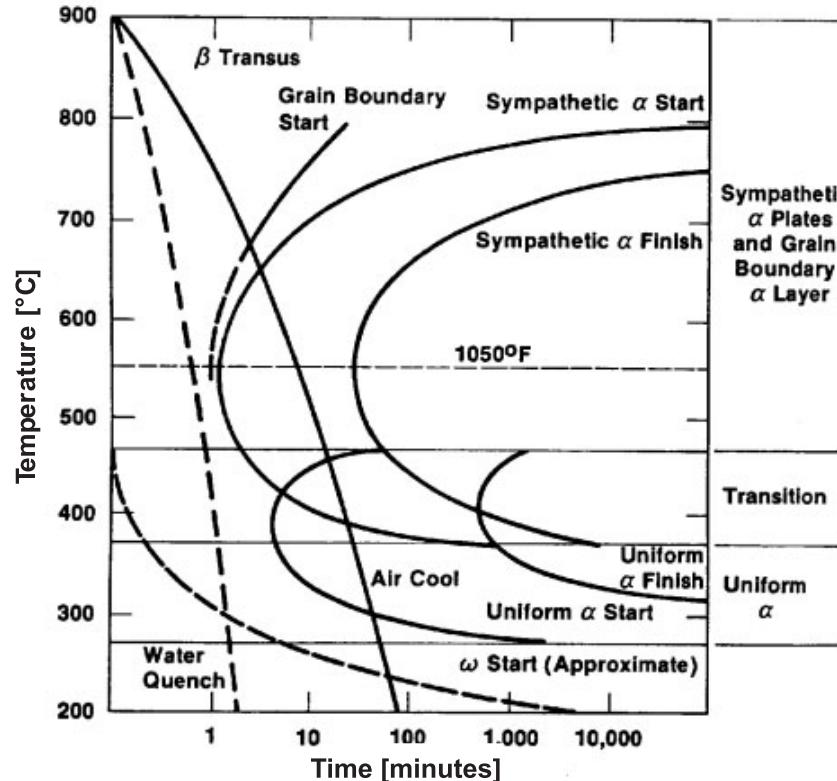
Beta titanium alloys are the most versatile class of titanium alloys. They offer the highest strength to weight ratios and very attractive combinations of strength, toughness, and fatigue resistance at large cross sections. Some of the disadvantages compared to  $\alpha+\beta$  alloys are increased density, a rather small processing window, and higher cost.

A beta alloy is a titanium alloy with sufficient  $\beta$ -stabilizer to suppress the martensitic transformation during quenching to room temperature.

The beta stability can be described by the molybdenum equivalent, which combines the effects of the various  $\beta$ -stabilizing elements like Mo, V, Fe, Cr, Nb, etc. A minimum value of about 10% is necessary to stabilize the beta phase during quenching

Tab. 2.2 Composition, category, applications, source and year of introduction of major beta titanium alloys [2].

<b>Alloy composition</b>	<b>Commercial name</b>	<b>Category</b>	<b>Mo-Eq u.</b>	<b>Actual and potential applications</b>	<b>Introduction Year-by</b>
Ti-35V-15Cr	Alloy C	beta	47	burn-resistant alloy	90-P & W
Ti-40Mo		beta	40	corrosion resistance	52-RemCru
Ti-30Mo		beta	30	corrosion resistance	52-RemCru
Ti-6V-6Mo-5.7Fe-2.7Al	TIMETAL 125	metastable	24	high-strength fasteners	90-TIMET
Ti-13V-11Cr-3Al	B 120 VCA	metastable	23	airframe, landing gear, springs	52-RemCu
Ti-1Al-8V-5Fe	1-8-5	metastable	19	fasteners	57-RMI
Ti-12Mo-6Zr-2Fe	TMZF	metastable	18	orthopedic implants	92-How-medica
Ti-4.5Fe-6.8Mo-1.5Al	TIMETAL LCB	metastable	18	low cost, high strength alloy	90-TIMET
Ti-15V-3Cr-1Mo-0.5Nb-3Al-3Sn-0.5Zr	VT 35	metastable	16	high strength airframe castings	na*-Russia
Ti-3Al-8V-6Cr-4Mo-4Zr	Beta C	metastable	16	oil fields, springs, fasteners	69-RMI
Ti-15Mo	IMI 205	metastable	15	corrosion resistance	58-IMI
Ti-8V-8Mo-2Fe-3Al	8-8-2-3	metastable	15	high strength forgings	69-TIMET
Ti-15Mo-2.6Nb-3Al-0.2Si	TIMETAL 21S	metastable	13	oxidation/corrosion resistant, TMCs	89-TIMET
Ti-15V-3Cr-3Sn-3Al	15-3	metastable	12	sheet, plate, airframe castings	78-USAF
Ti-11.5Mo-6Zr-4.5Sn	Beta III	metastable	12	high strength	69-Crucible
Ti-10V-2Fe-3Al	10-2-3	metastable	9.5	high strength forgings	71-TIMET
Ti-5V-5Mo-1Cr-1Fe-5Al	VT 22	metastable	8.0	high strength forgings	na*-Russia
Ti-5Al-2Sn-2Zr-4Mo-4Cr	Ti-17	beta-rich	5.4	high strength, medium temperature	68-GEAE
Ti-4.5Al-3V-2Mo-2Fe	SP 700	beta-rich	5.3	high strength, SPF	89-NKK
Ti-5Al-2Sn-2Cr-4Mo-4Zr-1Fe	Beta-CEZ	beta-rich	5.1	high strength, medium temperature	90-CEZUs
Ti-13Nb-13Zr		beta-rich	3.6	orthopedic implants	92-Smith & N.



Processing of  $\beta$  alloys usually consists of a hot working operation followed by a heat treatment. The final hot working step is normally performed in the  $\alpha+\beta$  field for the leaner  $\beta$  alloys, and preferentially in the  $\beta$  field for the richer beta alloys. The heat treatment consists of a solution treatment followed by quenching and a subsequent aging treatment.

Fig. 2.2 Qualitative TTT-diagram for Ti-10-2-3 ( $\beta$ -ST) [4, 5].

A solution heat treatment above the  $\beta$  -transus temperature results in coarse  $\beta$  grains. Solution treating slightly below the  $\beta$  transus leads to the precipitation of primary  $\alpha$  ( $\alpha_p$ ). The heat treatment temperature controls the  $\alpha_p$  volume fraction, while forging and rolling deformation influences the  $\alpha_p$  shape. Without working a needle-like ap shape develops; an increased amount of hot working leads to a globular ap shape.

At lower temperatures, typically 400 to 600 C, the secondary  $\alpha$  ( $\alpha_s$ ) precipitates in a fine distribution. It has a significant strengthening effect depending on its volume fraction and size, which in turn are controlled by aging temperature and time as well as by the solution treatment temperature. The precipitation of  $\alpha_s$  can be homogeneous as it is found in lean beta alloys like Ti-10-2-3, or inhomogeneous in richer beta alloys, like Beta C or Ti-15-3. Cold work generally enhances the aging response and can lead to a more homogenous distribution of the  $\alpha_s$ .

## Tensile Properties

Through aging, a wide range of yield stresses (typically 900 to 1400 MPa) can be reached in  $\beta$  titanium. With increased aging, however, all  $\beta$  alloys show a significant reduction in ductility. In more highly  $\beta$ -stabilized alloys like Ti-15-3 or Beta C, with an inhomogeneous as precipitation, duplex aging procedures have been developed. They consist of “high/low”- or “low/high” aging sequences. Duplex aging allows, for example, higher strength in shorter time than single-step aging. Duplex aging was mainly developed for an improvement of toughness and fatigue resistance.

Besides the dominating effect of aging, primary  $\alpha_p$  can also influence ductility.

**A coarsening of the  $\alpha_p$  as well as a change from globular to acicular leads to a reduction in ductility in Ti-10-2-3.**

## Fracture Toughness

Increased aging significantly reduces fracture toughness.

Fractography has revealed that as for ductility, an increased strain localization and increased strength difference between the soft  $\alpha_p$  and the aged matrix is the reason for this trend.

Duplex aging has been tried in order to increase the strength and toughness compared to single-step aging. *Results are mixed for different alloys....*

*For Ti 15-3  $K_{IC}$  from 43 to 66 MPa  $\sqrt{m}$  (with high/low aging) but for Ti 10-1-3 is better low/high....*

## Fatigue

The good fatigue potential of beta alloys is well known. A high cycle fatigue strength (HCF) around 700 MPa can be achieved in Ti-10-2-3 for large cross sections (> 100 mm); this is not possible for any other titanium alloy.

An increase in aging, or 0.2% yield strength, can increase HCF strength.

Richer beta alloys like Beta C or Ti-15-3 have a lower fatigue strength level than the leaner alloys such as Ti-10-2-3 or SP 700. One possible reason may be the trend for an inhomogeneous precipitation of as in the richer alloys. For Beta C crack nucleation occurs in the transgranular precipitate-free regions. Using a duplex aging procedure a more homogeneous as precipitation resulting in a 50 MPa increase in fatigue strength was achieved

A grain size reduction has been shown to increase the fatigue strength

## Applications

Despite the obvious potential of beta alloys, their share of the titanium market is still small (1% of the US market). However, their use is continuously increasing, especially in aerospace. This is through increased use of existing alloys like Ti-10-2-3. Most of the landing gear of the Boeing 777 airplane, for example, is produced from Ti-10-2-3

In addition, the rotor head of the Westland Super Lynx helicopter is also produced from Ti-10-2-3 instead of Ti-6-4.

The driving force for selection of Ti-10-2-3 is increased fatigue properties of the beta alloy.

Outside aerospace, beta alloys can be used for downhole service (deep oil and gas wells) where Beta C is a particularly appropriate candidate because of its good combination of mechanical and corrosion properties.

New alloys under development are: TIMETAL LCB (LCB – low cost beta) with its first application in automotive springs, SP 700 as a high strength alloy with improved cold and superplastic formabilit, and TMZF (Ti-12Mo-6Zr-2Fe) as a surgical implant alloy with low modulus, good strength, and corrosion resistance.