Geophysical Fluid Dynamics

Lecture V, VI, VII: Conservation Laws

- **Q** Leibniz Theorem for time derivative of volume integrals
- **2** Conservation of Mass Continuity Equation
	- Conservation Equation for a tracer
	- Advection-Diffusion
	- Diffusion
- **3** Conservation of Momentum
	- Cauchy
	- Navier-Stokes Equations
	- Euler Equation
	- The case of a rotating frame (towards the GFD Eq.)

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- **4** Conservation of Energy
	- Kinetic, Mechanical, Potential and Total Energy
	- First and Second law of thermodynamics
	- Bernoulli's Equation Principle

Geophysical Fluid Dynamics

Lecture VI: Conservation Laws

4 Conservation of Momentum

- Cauchy
- Navier-Stokes Equations
- Euler Equation
- The case of a rotating frame (towards the GFD Eq.)

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Conservation of Momentum: Forces in Fluids

- **1 BODY FORCES:** without physical contact. They exist because the medium is in a force field (gravitational, magnetic, electrostatic, electromagnetic, ...). We denote a body force by f.
- ² SURFACE FORCES: forces exerted on an area element by the surrounding through direct contact. They can be normal or tangential to the area (we know this).

$$
\tau_n \equiv \frac{dF_n}{dA}, \tau_s \equiv \frac{dF_s}{dA} \tag{1}
$$

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Conservation of Momentum

We simply apply Newton's law of motion to an infinitesimal fluid element. (The net force on the element must equal mass times the acceleration of the element)

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Conservation of Momentum

Considering the torque on a centroid axis and the motion for an infinitesimal fluid element, the sum of the Surface Forces in the x_1 direction then becomes $(\tau_{11}+\frac{\partial\, \tau_{11}}{\partial \varkappa_1}$ $\frac{dx_1}{2} - \tau_{11} + \frac{\partial \tau_{11}}{\partial x_1}$ $\frac{dx_1}{2}$)dx₂dx₃ + (...)dx₁dx₃ + (...)dx₁dx₂. Reducing to

$$
\left[\left(\frac{\partial \tau_{11}}{\partial x_1} + \frac{\partial \tau_{21}}{\partial x_2} + \frac{\partial \tau_{31}}{\partial x_3} \right) dx_1 dx_2 dx_3 = \frac{\partial \tau_{j1}}{\partial x_j} dV \right]
$$
 (2)

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Conservation of Momentum: Cauchy's Equation

$$
\left(\frac{\partial \tau_{11}}{\partial x_1} + \frac{\partial \tau_{21}}{\partial x_2} + \frac{\partial \tau_{31}}{\partial x_3}\right) dx_1 dx_2 dx_3 = \frac{\partial \tau_{j1}}{\partial x_j} dV \tag{3}
$$

So generalizing, the i-component of the SURFACE FORCE per unit volume is $\frac{\partial \tau_{ij}}{\partial x_j}$ [We can prove that the stress tensor is symmetric, so that $\tau_{ii} = \tau_{ii}$] Newton's law gives

$$
\rho \frac{Du_i}{Dt} = f_i + \frac{\partial \tau_{ij}}{\partial x_j} \tag{4}
$$

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Conservation of Momentum: Cauchy's Equation

Equation of motion relating acceleration to the net force at a point and holds for any continuum, solid or fluid, no matter how the stress tensor τ_{ii} is related to the deformation field.

$$
\overline{\rho \frac{Du_i}{Dt}} = f_i + \frac{\partial \tau_{ij}}{\partial x_j} \tag{5}
$$

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where f is a BODY FORCE per unit mass (like Newtonian gravity), so that ρf is the body force per unit volume. This is the CAUCHY'S EQUATION.

A relation between stress and deformation in a continuum is called a constitutive equation. It linearly relates the stress to the rate of strain in a fluid medium.

In a fluid at rest, there are only normal forces / stresses. And we know that the stress tensor is isotropic. A second-order isotropic tensor is the Kronecker delta δ . Any isotropic tensor must be proportional to δ . Therefore, in static, the stress must take the form

$$
\tau_{ij} = -\rho \delta_{ij},\tag{6}
$$

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where p is thermodynamic pressure, $f(\rho, T)$. The negative sign is because the normal components of τ are considered positive if they indicate tension, rather than compression.

A **moving fluid** develops more components of stress due to viscosity, and shear stresses develop. The diagonal terms of τ_{ii} are now unequal.

$$
\tau_{ij} = -\rho \delta_{ij} + \sigma_{ij}.\tag{7}
$$

 σ_{ij} is the deviatoric stress tensor, nonisotropic, related to the velocity gradient tensor $\partial \, u_i/\partial x_j$:

$$
\partial u_i/\partial x_j = e_{ij} + r_{ij} \tag{8}
$$

The rotation tensor, the antisymmetric part, can not generate stress as it represents fluid rotation and not deformation. Stresses are generated by the strain rate tensor e_{ij} .

$$
e_{ij} \equiv \frac{1}{2} \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) \tag{9}
$$

... Now a few assumptions (and jumps) ...

• Hypothesis of Newtonian Fluid: We assume a linear relationship between shear stresses and deformations $\tau = \mu (du/dy)$, so that we have

$$
\sigma_{ij} = K_{ijmn} e_{mn}.\tag{10}
$$

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 K_{jimn} is a 4th order tensor (81 components) that depends on the thermodynamic state of the medium. This says that each stress component is linearly related to all nine component of $e_{ii} \rightarrow 81$ constants are needed for this.

- Hypothesis of isotropic medium: from 81 coefficients of K_{ijmn} we go down to 3!
- Hypothesis of symmetric σ_{ii} : from 3 coefficients of K_{iimn} we go down to 2!
- Stokes' Hypothesis: pressure p of the fluid is equal to the thermodynamic pressure or to that of the static fluid, \hat{p} . From 2 coefficients of K_{iimn} we go down to 1!

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Constitutive Equation for Newtonian Fluids So from here:

$$
\tau_{ij}=-\rho\delta_{ij}+\sigma_{ij}.
$$

we get to the final form of the constitutive equation, which is

$$
\boxed{\tau_{ij} = -(\rho + \frac{2}{3}\mu \nabla \cdot \mathbf{u})\delta_{ij} + 2\mu e_{ij}}
$$
\n(11)

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- The linear relation between τ and **e** is consistent with Newton's definition of viscosity coefficient in a parallel flow $u(y)$, so that Eq.([??](#page-11-0)) gives a shear stress of $\tau = \mu(du/dy)$. Hence, this only applies to Newtonian fluids.
- The nondiagonal terms of Eq. ([??](#page-11-0)) relate the shear stress to the shear strain rate

$$
\tau_{12} = \mu \left(\frac{\partial u_1}{\partial x_2} + \frac{\partial u_2}{\partial x_1} \right)
$$

Now let's plug the constitutive equation into the Cauchy's equation to get the Equation of Motion for a Newtonian Fluid

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$$
\rho \frac{Du_i}{Dt} = f_i + \frac{\partial \tau_{ij}}{\partial x_j}
$$

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and the constitutive Eq.:

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\tau_{ij} = -(\rho + \frac{2}{3}\mu \nabla \cdot \mathbf{u})\delta_{ij} + 2\mu e_{ij}
$$

The two give the general form of the Navier-Stokes equation:

$$
\left[\rho \frac{Du_i}{Dt} = f_i - \frac{\partial \rho}{\partial x_i} + \frac{\partial}{\partial x_j} \left[2\mu e_{ij} - 2/3\mu (\nabla \cdot \mathbf{u}) \delta_{ij}\right]\right]
$$

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$$
\rho \frac{Du_i}{Dt} = f_i - \frac{\partial p}{\partial x_i} + \frac{\partial}{\partial x_j} \Big[2\mu e_{ij} - 2/3\mu (\nabla \cdot \mathbf{u}) \delta_{ij} \Big]
$$

we have noted that $(\partial p/\partial x_i)\delta_{ii} = (\partial p/\partial x_i)$.

Viscosity μ is generally a function of the thermodynamic state. For fluids, generally μ depends strongly on Temperature:

$$
\begin{cases} \mu & \text{decreases as T increases} \\ \mu & \text{increases as T increases} \\ \end{cases}
$$
 for Gases

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- If temperature differences are small enough within the fluid, then $\mu \sim$ constant.
- Also, for incompressible fluids: $\nabla \cdot \mathbf{u} = 0$, and so we can write:

$$
\rho \frac{Du_i}{Dt} = f_i - \frac{\partial \rho}{\partial x_i} + 2\mu \frac{\partial}{\partial x_j} e_{ij}
$$

which is

$$
\rho \frac{Du_i}{Dt} = f_i - \frac{\partial p}{\partial x_i} + \mu \nabla^2 u_i
$$

or

$$
\rho \frac{Du}{Dt} = \mathbf{f} - \nabla p + \mu \nabla^2 \mathbf{u}
$$

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If we neglect viscous effects , which is true far from boundaries, we obtain the Euler Equation:

$$
\rho \frac{Du_i}{Dt} = f_i - \frac{\partial \rho}{\partial x_i}
$$

or

$$
\boxed{\rho \frac{D\mathbf{u}}{Dt} = \mathbf{f} - \nabla p}
$$

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Body forces on a Rotating frame

- **1** In a non-rotating frame: $f = g$
- **2** In a rotating frame: $f = g + \{ \text{apparent force due to rotation} \}$

It can be easily proved that, in a rotating frame of reference (such as the Earth), the velocities between a fixed frame of reference and a frame of reference rotating at a uniform angular velocity Ω are related by:

$$
\mathbf{u}_{\mathbf{f}} = \mathbf{u}_{\mathbf{r}} + \mathbf{\Omega} \times \mathbf{r}, \tag{12}
$$

where **r** is the position vector. This means that, after some manipulation (see any textbook ...),

$$
a_f = a + 2\Omega \times u.
$$
 (13)

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So that the (inertial) acceleration equals the acceleration in a rotating system plus the Coriolis acceleration.

The Coriolis force / acceleration / effect

We now have

$$
\frac{D\mathbf{u}}{Dt} + 2\Omega \times \mathbf{u} = \mathbf{g} - \frac{1}{\rho} \nabla \rho + v \nabla^2 u_{ij},
$$

where $v = \mu/\rho$ is the kinematic viscosity. The earth rotates at a rate

$$
\Omega=2\pi\,\text{rad}/\text{day}=0.73\times10^{-4}\text{s}^{-1}
$$

We decompose the components of the angular velocity of the earth:

$$
2\Omega \times \mathbf{u} = 2\Big[\big(\Omega_{\mathbf{y}} w - \Omega_{\mathbf{z}} v \big) \hat{i} + \big(\Omega_{\mathbf{z}} u - \Omega_{\mathbf{x}} w \big) \hat{j} + \big(\Omega_{\mathbf{x}} v - \Omega_{\mathbf{y}} u \big) \hat{k} \Big] \tag{14}
$$

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Thin layer approximation on a rotating sphere

For atmosphere and oceans the depth scale of flow \sim few kilometers. The horizontal scale is instead \sim hundreds/thousands of kilometeres.

So we can make the *thin layer approximation*: $|w| < u$. We can understand this from the continuity equation.

So the Coriolis components reduce to:

$$
2\Omega \times \mathbf{u} = 2\Big[-\Omega_z v \hat{i} + \Omega_z u \hat{j} + (\Omega_x v - \Omega_y u) \hat{k} \Big].
$$
 (15)

Note that:

$$
\Omega_z = \Omega \sin \theta
$$

$$
\Omega_y = \Omega \cos \theta
$$

$$
\mathbf{g} = -g\hat{k}
$$

and let's define $f = 2\Omega sin\theta$ (twice the vertical component of Ω). **KORK STRATER STRAKER**

The Coriolis force / acceleration / effect

Planetary Vorticity (or Coriolis parameter, or Coriolis frequency):

$$
f = 2\Omega \sin \theta \tag{16}
$$

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f is max at the poles (max spin) and zero at the Equator (only translation).

The Equations of motion on a rotating earth

$$
2\Omega \times \mathbf{u} = 2 \Big[-f\mathbf{v}\,\hat{\mathbf{i}} + f u \,\hat{\mathbf{j}} - \Omega \sin \theta \, u \,\hat{\mathbf{k}} \Big]. \tag{17}
$$

$$
\frac{Du}{Dt} - f_V = -\frac{1}{\rho} \frac{\partial P}{\partial x} + v \nabla^2 u \qquad (18)
$$

$$
\frac{Dv}{Dt} + fu = -\frac{1}{\rho} \frac{\partial P}{\partial y} + v \nabla^2 v \qquad (19)
$$

$$
\frac{Dw}{Dt} - (2\Omega cos\theta)u = -\frac{1}{\rho}\frac{\partial P}{\partial z} + v\nabla^2 w - g \qquad (20)
$$

but the vertical component of the Coriolis force is negligible compared to the dominant terms in the vertical equation of motion

The Equations of motion on a rotating earth

$$
\frac{Du}{Dt} - f_V = -\frac{1}{\rho} \frac{\partial P}{\partial x} + v \nabla^2 u \qquad (21)
$$

\n
$$
\frac{Dv}{Dt} + fu = -\frac{1}{\rho} \frac{\partial P}{\partial y} + v \nabla^2 v \qquad (22)
$$

\n
$$
\frac{Dw}{Dt} = -\frac{1}{\rho} \frac{\partial P}{\partial z} + v \nabla^2 w - g \qquad (23)
$$

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The equation of motion for a thin shell on a rotating earth. Only the vertical component of the earth's angular velocity appears as a consequence of the flatness of the fluid trajectories.

Conservation laws in fluid mechanics

$$
\frac{Du}{Dt} - f_V = -\frac{1}{\rho} \frac{\partial P}{\partial x} + v \nabla^2 u \qquad (24)
$$

\n
$$
\frac{Dv}{Dt} + fu = -\frac{1}{\rho} \frac{\partial P}{\partial y} + v \nabla^2 v \qquad (25)
$$

\n
$$
\frac{Dw}{Dt} = -\frac{1}{\rho} \frac{\partial P}{\partial z} + v \nabla^2 w - g \qquad (26)
$$

\n
$$
\frac{D\rho}{Dt} + \rho \nabla \cdot u = 0 \qquad (27)
$$

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Next, conservation of Energy and Bernoulli's Equation.