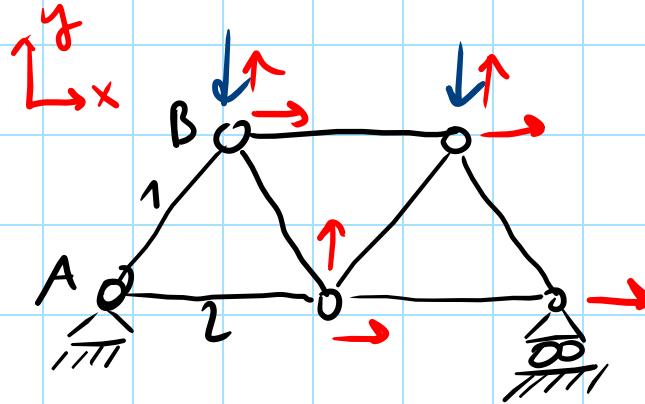


25/03/25

TRUSS STRUCTURES: SOLUTION BASED ON "METHOD OF DISPLACEMENTS"



NODAL DISPLACEMENTS ARE NOW THE UNKNOWNs

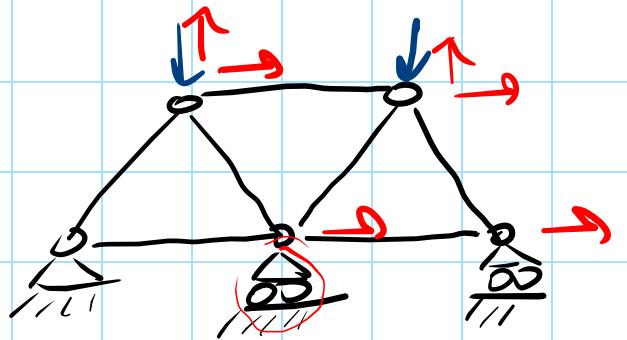
(7 IN OUR EXERCISE). IN ORDER TO SOLVE THE PROBLEM, WE

CAN WRITE A SYSTEM OF THE 7 EQUIL. EQS ASSOCIATED WITH
THE DIRECTIONS OF THE UNKNOWNs. TO ELIMINATE FORCES OF
BARS WE INTRODUCE THE ELASTIC CONSTITUTIVE LAW ($N_i = f(u_j)$)

\Rightarrow SYSTEM AS A FUNCTION OF THE 7 DISPLACEMENTS.

THERE ARE STILL 3 EQUIL. EQS THAT ALLOW THE CALCULATION OF
THE 3 EXTERNAL REACTIONS.

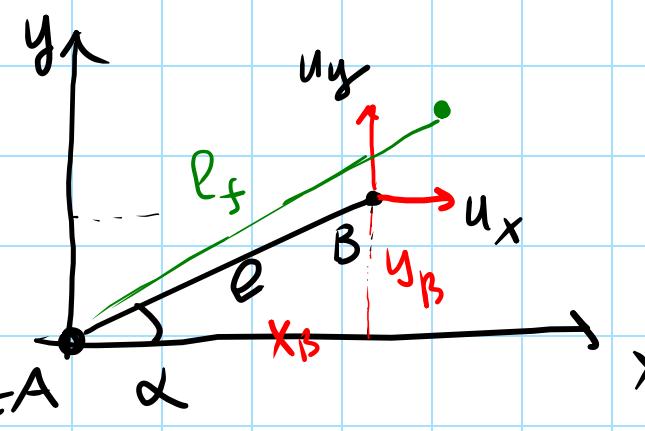
THE ADVANTAGE OF THE METHOD IS THAT IT CAN BE APPLIED IN THE SAME WAY
FOR A REDUNDANT STRUCTURE:



6 UNKNOWNs \Rightarrow 6 EQUIL. EQS + CONSTITUTIVE LAW
LATER

4 EXTERNAL WITH THE REMAINING 4 EQUIL. EQS
REACTIONS

INTRODUCTION OF DISPLACEMENT METHODS FOR TRUSS STRUCTURES

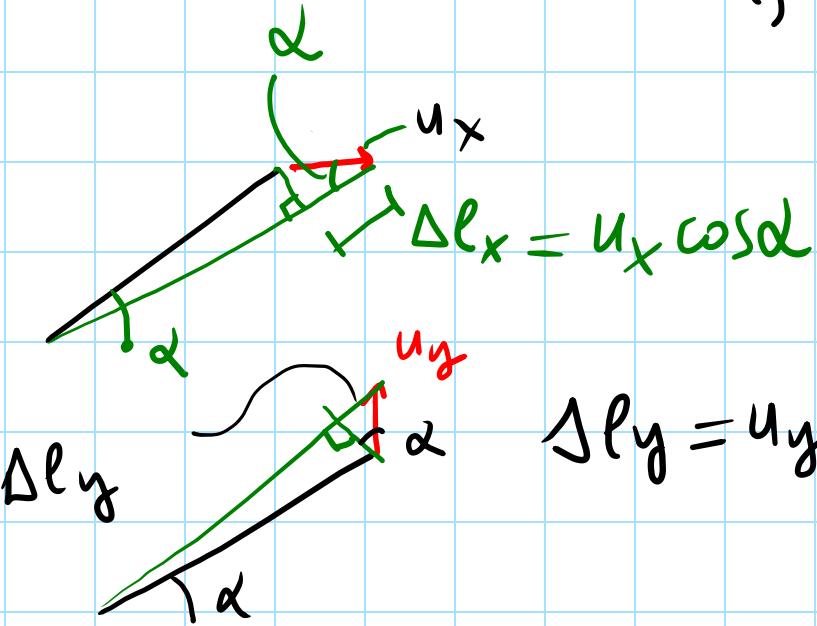


$$l = \sqrt{x_B^2 + y_B^2} ; l_f = \sqrt{(x_B + u_x)^2 + (y_B + u_y)^2} = l_f(u_x, u_y)$$

EXPAND l_f TO THE FIRST ORDER IN u_x AND u_y

$$l_f = l_f(0,0) + \frac{\partial l_f}{\partial u_x} \Big|_{(0,0)} u_x + \frac{\partial l_f}{\partial u_y} \Big|_{(0,0)} u_y + O(\sqrt{u_x^2 + u_y^2})$$

$$\Delta l = l_f - l \\ = \Delta l_x + \Delta l_y$$



$$\Delta l_y = u_y \sin \alpha$$

RECALL THAT THE CONSTITUTIVE EQUATION FOR A BAR IS

$$N = \frac{EA}{l} \Delta l$$

$$l_f \approx l + \left[\frac{1}{2\sqrt{2}} [2(x_B + u_x)] \right]_{(0,0)} u_x + \left[\frac{1}{2\sqrt{2}} [2(y_B + u_y)] \right]_{(0,0)} u_y = l + \frac{x_B}{l} u_x + \frac{y_B}{l} u_y$$

$$\cos \alpha \quad \sin \alpha \\ \Delta l_x \quad \Delta l_y$$

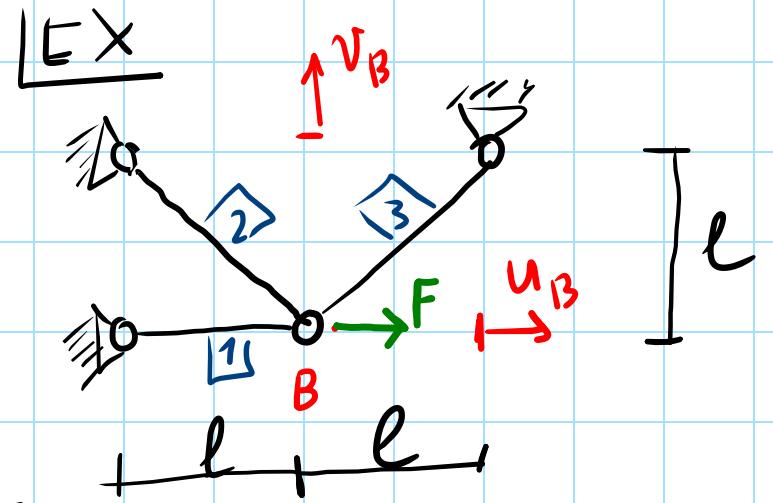
$$\sigma = E \epsilon$$

$$\frac{N}{A} = E \frac{\Delta l}{l}$$

$$N = \frac{EA}{l} \Delta l$$

K: EQUIV.
SPRING
STIFFNESS

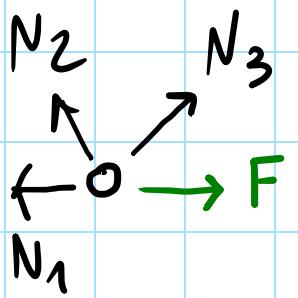




E: const

$$A_1 = A; A_2 = A_3 = \sqrt{2}A$$

$$l_1 = l; l_2 = l_3 = \sqrt{2}l$$



2 EQUIL. EQS

$$\begin{aligned} \xrightarrow{\text{+}}: -N_1 - N_2 \frac{1}{\sqrt{2}} + N_3 \frac{1}{\sqrt{2}} + F &= 0 \\ +\uparrow: N_2 \frac{1}{\sqrt{2}} + N_3 \frac{1}{\sqrt{2}} &= 0 \end{aligned}$$

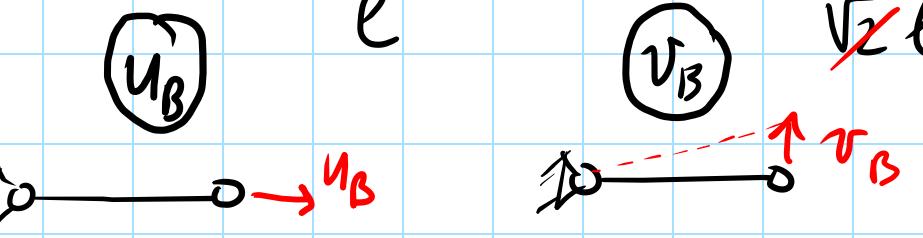
u_B, v_B : UNKNOWN S

CONSTITUTIVE EQS: $N_i = N_i(u_B, v_B)$

$$N_1 = \frac{EA}{l} \Delta l_1; N_2 = \frac{E\sqrt{2}A}{\sqrt{2}l} \Delta l_2; N_3 = \frac{E\sqrt{2}A}{\sqrt{2}l} \Delta l_3$$

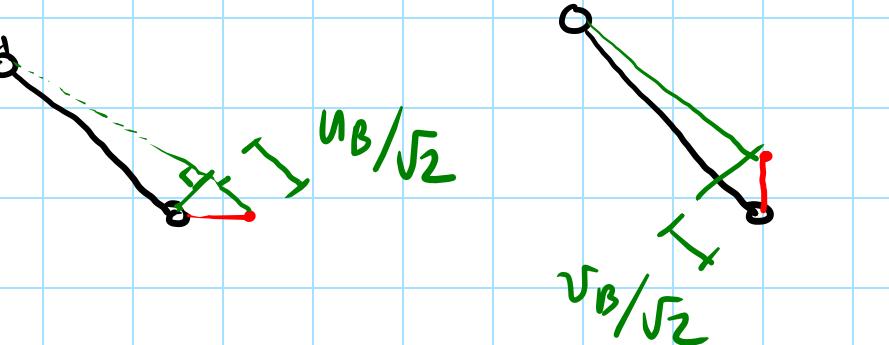
$$K = EA/l$$

Δl_1)



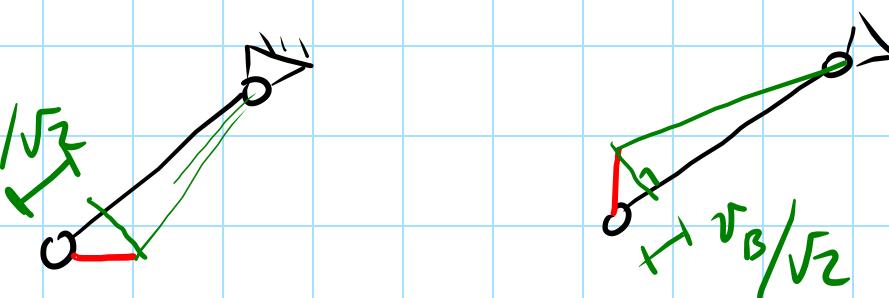
$$\Delta l_1 = +u_B$$

Δl_2)



$$\Delta l_2 = +\frac{u_B}{\sqrt{2}} - \frac{v_B}{\sqrt{2}}$$

Δl_3)



$$\Delta l_3 = -\frac{u_B}{\sqrt{2}} - \frac{v_B}{\sqrt{2}}$$

$$\left. \begin{aligned} -K u_B - \frac{K}{\sqrt{2}\sqrt{2}} (u_B - v_B) + \frac{K}{\sqrt{2}\sqrt{2}} (-u_B - v_B) + F &= 0 \\ \frac{K}{2} (u_B - v_B) + \frac{K}{2} (-u_B - v_B) &= 0 \end{aligned} \right\}$$

$$-\frac{K}{2} u_B + \frac{K}{2} (u_B - v_B) + \frac{K}{2} (u_B + v_B) = -F$$

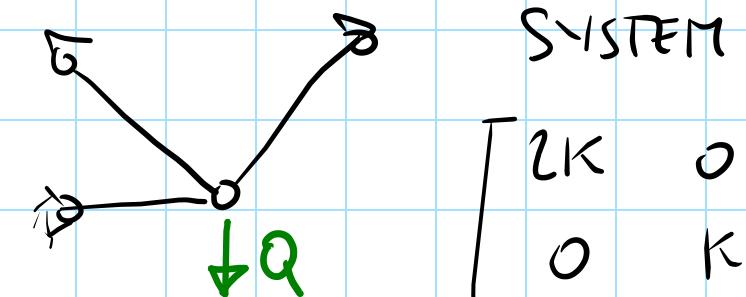
$$\frac{K}{2} (u_B - v_B) - \frac{K}{2} (u_B + v_B) = 0$$

$$N_1 = K u_B = F/2$$

$$N_2 = \frac{K}{\sqrt{2}} (u_B - v_B) = \frac{F}{2\sqrt{2}}$$

$$N_3 = -\frac{K}{\sqrt{2}} (u_B + v_B) = -\frac{F}{2\sqrt{2}}$$

COMMENT : ①



$$\begin{bmatrix} 2K & 0 \\ 0 & K \end{bmatrix} \begin{bmatrix} u_B \\ v_B \end{bmatrix} = \begin{bmatrix} 0 \\ -Q \end{bmatrix}$$

SOLUTION

$$u_B = 0$$

$$v_B = -\frac{Q}{K}$$

$$(N_1 = 0)$$

$$\begin{bmatrix} 2K & 0 \\ 0 & -K \end{bmatrix} \begin{bmatrix} u_B \\ v_B \end{bmatrix} = \begin{bmatrix} -F \\ 0 \end{bmatrix} ; \quad u_B = \frac{F}{2K}, \quad v_B = 0$$

↓

$$\begin{bmatrix} 2K & 0 \\ 0 & K \end{bmatrix} \begin{bmatrix} u_B \\ v_B \end{bmatrix} = \begin{bmatrix} +F \\ 0 \end{bmatrix}$$

STIFFNESS MATRIX (SYMMETRIC AND
[K] POSITIVE DEFINITE)

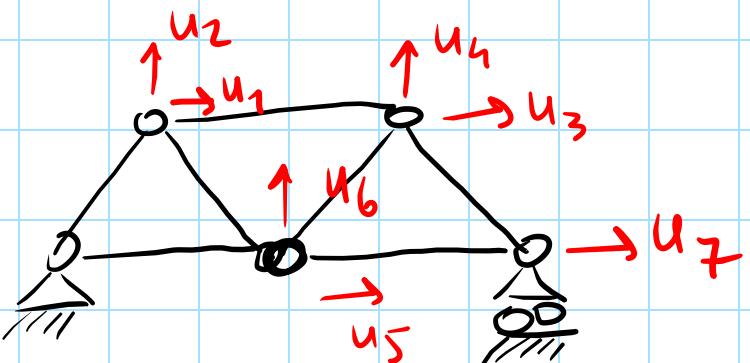
② MEANING OF K_{11}

first line of the system : $K_{11} u_B + K_{12} v_B = F$

TAKE $u_B = 1$ AND $v_B = 0 \Rightarrow K_{11} = F$

TAKE $u_B = 0$ AND $v_B = 1 \Rightarrow K_{12} = F$

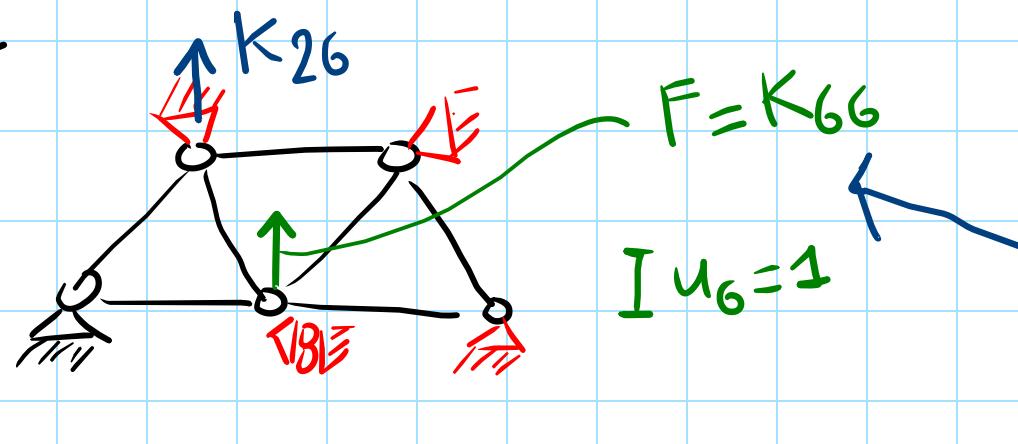
K_{ij} IS THE "FORCE" ASSOCIATED WITH DOF i WHEN DOF $j=1$ AND THE
REMAINING DOFS ARE ALL NULL.



$$K_{66} ?$$

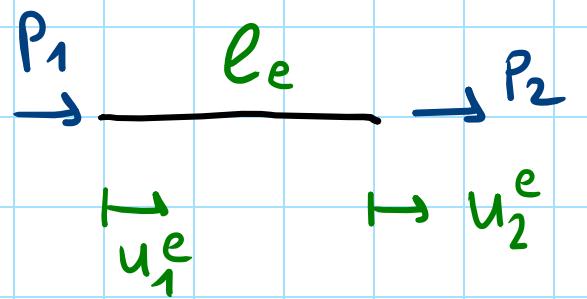
$$u_6 = 1$$

$$u_i \ (i \neq 6) = 0$$



K_{26} : REACTION ALONG DIRECTION 2 OF THE WADEN SYSTEM

LOCAL STIFFNESS MATRIX OF A BAR



$$u_2^e = 1$$

$$u_1^e = 0$$

$$K_{12} = -\frac{EA}{l_e} \cdot 1$$

$$K_{22} = \frac{EA}{l_e} \cdot 1$$

$$\begin{bmatrix} P_1 \\ P_2 \end{bmatrix} = \begin{bmatrix} K_{11} & K_{12} \\ K_{21} & K_{22} \end{bmatrix} \begin{bmatrix} u_1^e \\ u_2^e \end{bmatrix}$$

$$= \frac{EA}{l_e} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$

LOCAL STIFFNESS
MATRIX \underline{K}_e

$$u_1^e = 1$$

$$u_2^e = 0$$

$$K_{11} = \frac{EA}{l_e} \cdot 1$$

$$K_{21} = -\frac{EA}{l_e} \cdot 1$$